

# United States Patent [19]

Biersach

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## [54] HONEYCOMB STRUCTURE AND METHOD OF MAKING SAME

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[58] Field of Search ..... 156/197, 292, 153; 428/73, 116, 117, 118

## [56] References Cited

### U.S. PATENT DOCUMENTS

2,477,852 8/1949 Bacon ..... 428/116  
2,839,442 6/1958 Whitaker ..... 428/116 X  
3,007,834 11/1961 Moeller et al. .... 428/116 X  
3,483,070 12/1969 Kennedy et al. .... 428/118

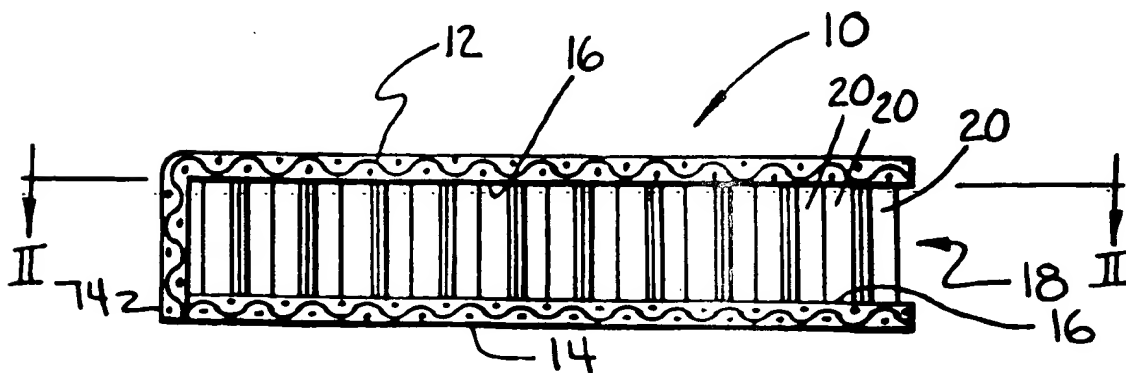
Primary Examiner—Henry F. Epstein

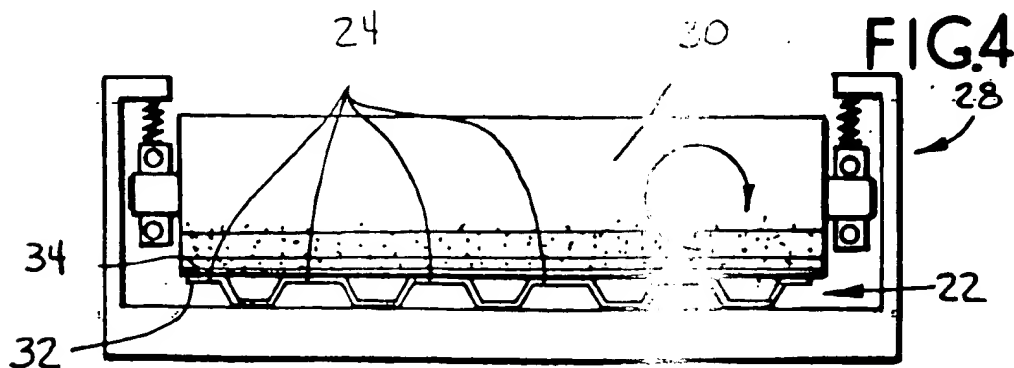
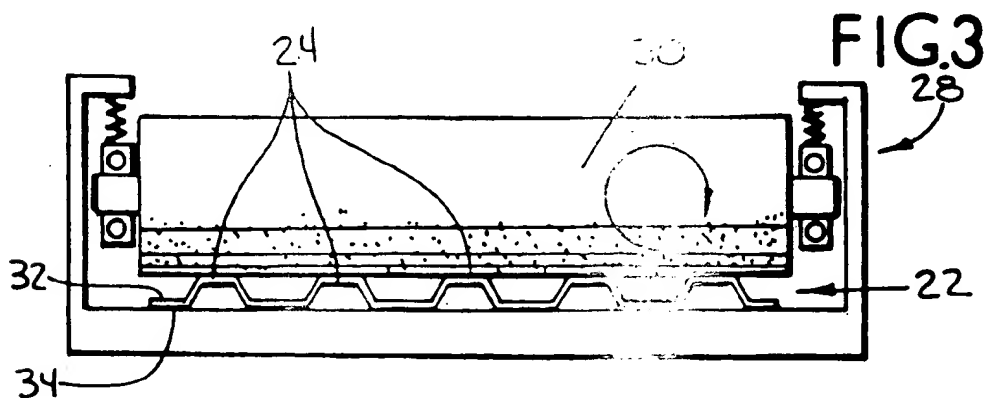
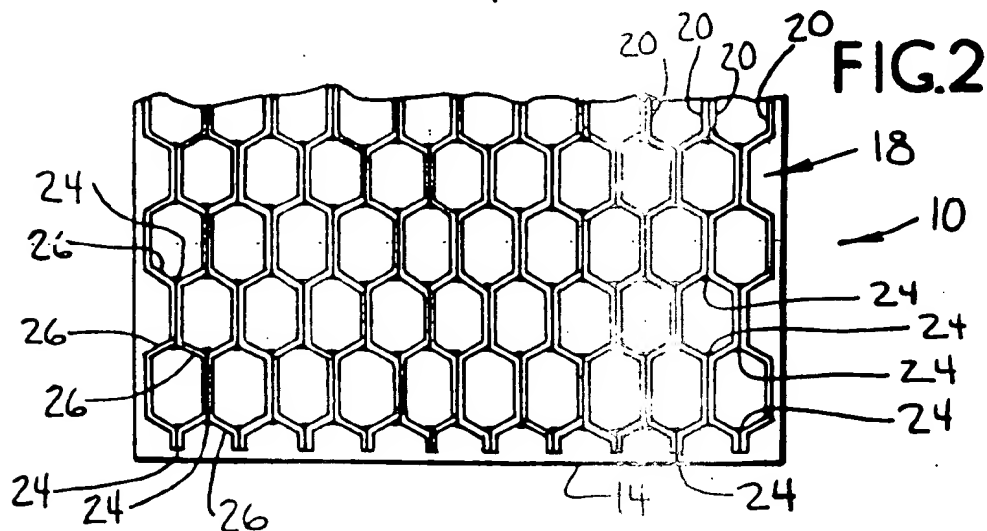
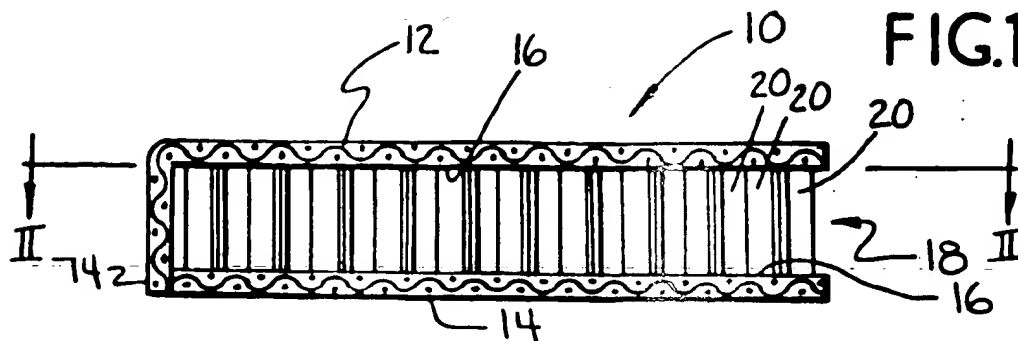
Attorney, Agent, or Firm—Henry C. Kovar

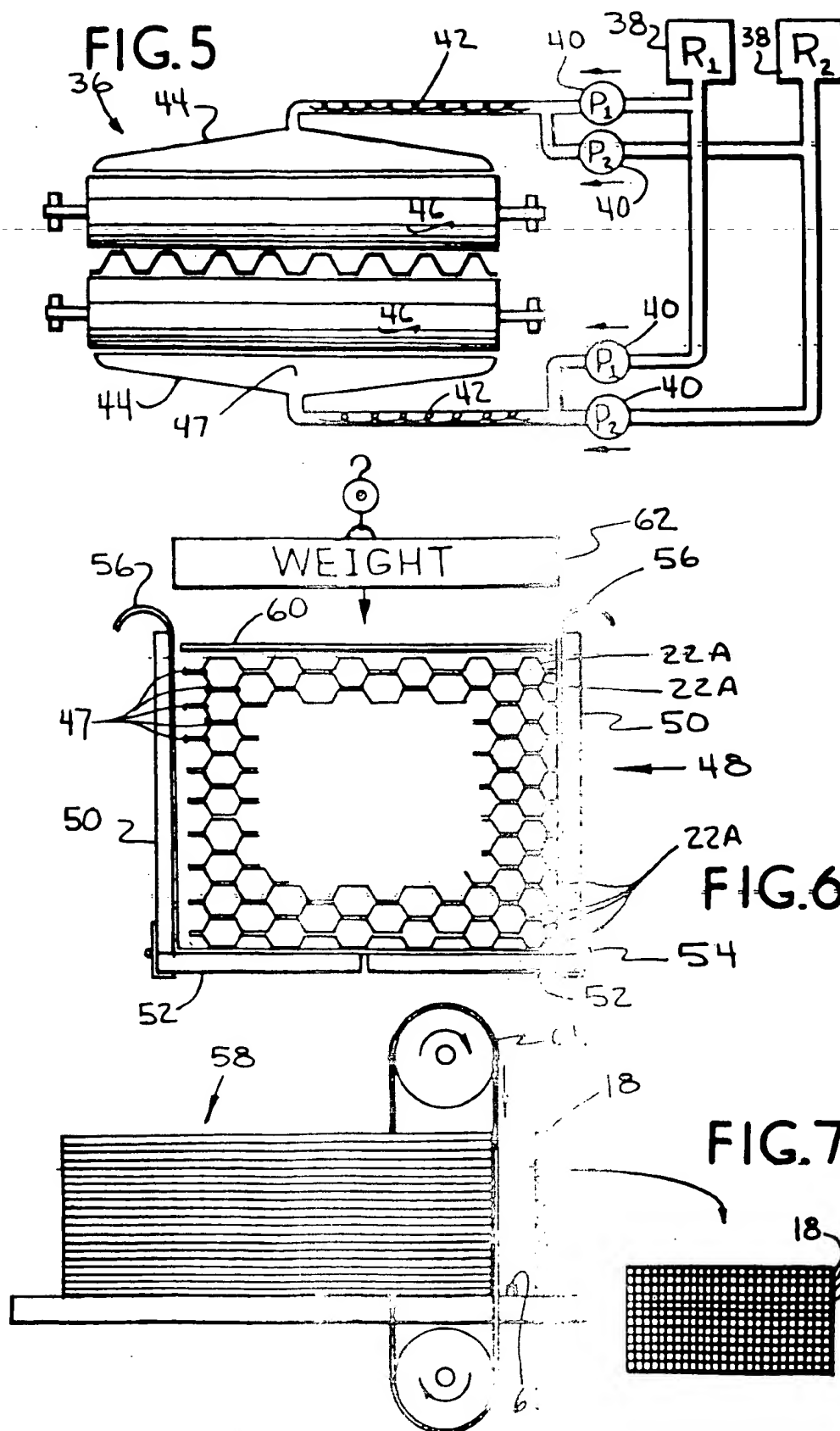
## [57] ABSTRACT

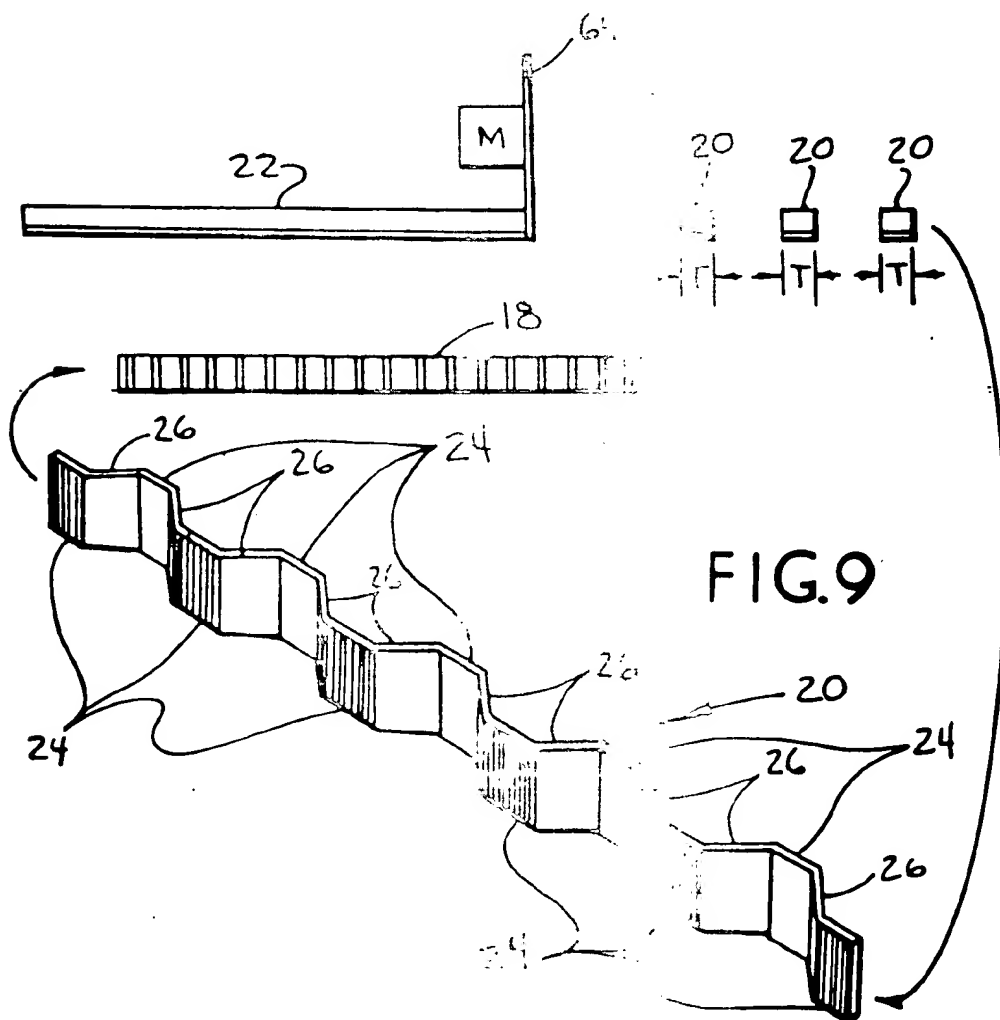
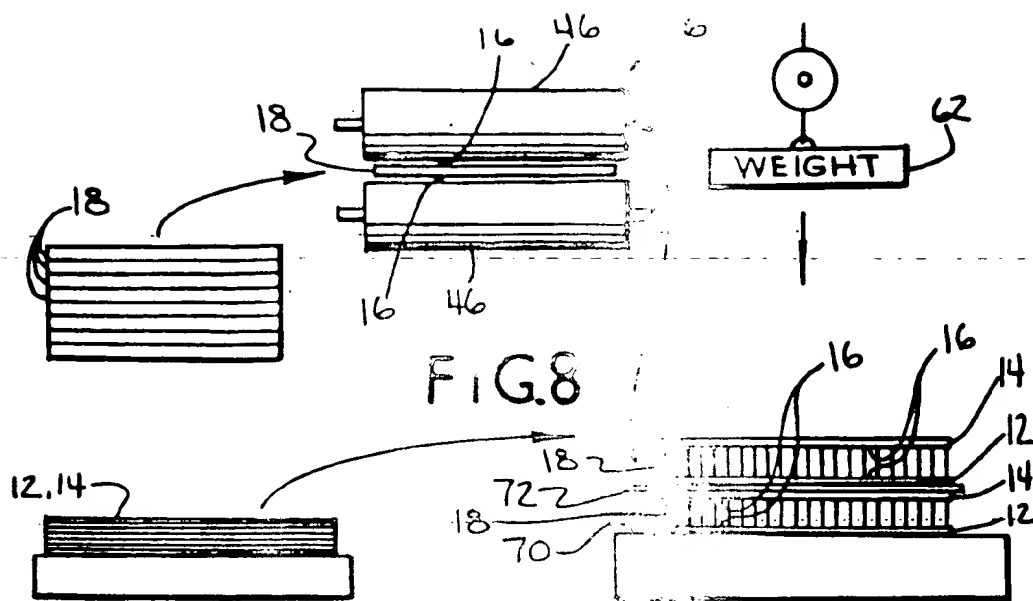
A new and improved honeycomb core and honeycomb panel are provided with a new method of making honeycomb cores and panels. The honeycomb core is fabricated from conventional or higher quality corrugated fiberglass sheet and the honeycomb panel has a resin core that is fiber filled for strength. The method of making the core sheet has the steps of providing corrugated rigid resin panels, abraiding the corrugation apexes on both sides of the corrugated panels, applying adhesive on the abraided apexes, stacking the corrugated panels with every other corrugated panel being flipped over so that the apexes face each other, bonding the apexes and therefore the corrugated panels together to form a core block, and then sawing desired thicknesses of core sheets off of the core blocks. The core sheets are subsequently bonded into finished honeycomb panels.

12 Claims, 3 Drawing Sheets









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FIG. 8 is a process flow diagram of the assembly of the completed core panels; and

FIG. 9 is an alternative process flow diagram.

### DESCRIPTION OF THE PREFERRED EMBODIMENT

In accordance with the principles of the present invention, the preferred embodiment of the new and improved honeycomb panel of the present invention is shown in FIGS. 1 and 2 and generally designated by the numeral 10, and hereinafter referred to as the panel 10 for brevity.

The panel 10 has at least one and usually two outer surface sheets 12, 14 which are spaced apart from each other and which are of a conventional material such as wood, fiber resin, metal, ceramic and so on. The inner surface of each surface sheet 12, 14 has a layer of appropriate adhesive 16 which is preferably an irreversible two-part reactive structural adhesive such as an epoxy resin. Within the surface sheets 12, 14 and the adhesive 16 is a new honeycomb core sheet 18 which is an important part of this invention. The core sheet 18 is permanently bonded to the surface sheets 12, 14 with and by the adhesive layers 16.

The core sheet 18 is comprised of a plurality of ribbons 20 of rigid corrugated resin panel pieces, which preferably are fiber filled for structural reinforcement. These ribbons 20 are cut either singularly or in bonded pluralities from one or more rigid corrugated resin panels 22 as will be explained.

Each panel 22 as shown in FIGS. 3 and 4 is a panel of nominal width and length, for example 4 feet wide and 12 feet long. The panel 22 has a plurality of corrugations that run the length of the panel 22 and which are evenly uniformly spaced from each other across the transverse width. On each side of each panel 22 and therefore each ribbon 20 is a plurality of corrugation apexes 24 which define the outermost surfaces on each side of the panel. The apexes 24 are preferably flats, and each apex 24 has a transverse flat dimension which is at least  $\frac{1}{2}$  the transverse width of a complete corrugation wave. The distance between opposing apexes 24 are thickness legs 26. The panels 22 of conventional build are commonly found in lumber yards and are used for the conventional construction of buildings, roofs, walls, shelters, and so forth. Be it understood that high cost and high performance aerospace resins and fibers are also used as material for the panels 22 and ribbons 20 and the various extrusion and/or formed profiles of the panels 22 and ribbons 20 are usable in this invention. However, the invention is particularly cost effective with fiber-glass reinforced polyester resin panels.

In the method of making the ribbons 20 and the core sheet 18, the entire panel 22 is firstly prepared by running a honeycomb material by being run through a sander 28 which has an abrasive wheel or belt 30 that abrades the entire surface of each and every corrugation apex 24 on a first side 32 of the panel 22. The panel 22 is then flipped over and again run through the sander 28 which abrades entirely, each and every surface of every corrugation apex 24 on the second side 34 of the panel 22. On high production facilities, it is recognized that a double wheel sanding machine can abraid both panel sides 32, 34 at one time and during a single pass.

The abraded panel 22A is then run through an adhesive application machine 36 as shown in FIG. 5 which has vats 38 for adhesive parts 1 and 2, adhesive pumps 40, adhesive mixers 42, adhesive spreaders 44 and applica-

tion rollers 46. Alternatively the core adhesive 47 can be applied by hand with rollers or brushes. Regardless, core adhesive 47 is applied to all the prepared apex surfaces 24 of at least one panel side 32 and also preferably on the apex surfaces 24 of the second panel side 34. The prepared panels 22A with the core adhesive 47 thereon are then transferred to the honeycomb core jig 48.

The core jig 48 as shown in FIG. 6 is a box having sides 50, ends 52 (shown) and single or double bottom doors 54 (not shown). The doors 52 lock in the closed position and a disposable box liner 56 is used for each box for easy lock-fabrication. The preferred liner 56 is a polyethylene film. The jig 48 is sized to be a reasonably close fit for the panels 22 so that the panels 22 align themselves snugly upon another. A plurality of prepared abraded panels 22A are placed into the jig 48 as shown in FIG. 7. Each adhesive panel 22A is flipped or inverted, for example the first and bottom panel 22A is placed with its first side 32 down and a second side 34 up. Each subsequent panel 22A will have its first side 32 up and its second side 34 down. All of the odd numbered panels 22A, i.e., panels 1, 3, 5, 7, 9, 11, and so forth (and so forth for the bottom) will be like the bottom panel 22A. The even numbered panels 22A (i.e., panels 2, 4, 6, 8, 10, 12 and so forth) are all positioned like the top panel 22A. The total number of panels 22A in the jig 48 determines the height of the core sheet 18. When the desired quantity of prepared abraded panels 22A have been placed in the jig 48, the doors 52 are closed and the stack is placed upon the top panel 22A. A box lid 58 is then placed atop the stack of panels 22A. The adhesive 47 on the panels 22A is then cured whereupon the adhesive 47 is irreversibly sets and the individual panels 22A are irreversibly structurally adhesively bonded together. The doors 52 are then opened and the core block 58 is removed. Alternatively, the core block 58 may be removed by pulling the liner 56 up and out of the box 58 in the line 56.

The core block 58 is then taken to a bandsaw 64, as shown in FIG. 8, and an adjustable stop 66 is set to produce a desired thickness of core sheet 18 and, the core sheets 18 are cut off of the stack as shown. The core sheets 18 are sawed with parallel sides and are of uniform thickness, however, variable thicknesses may be cut if desired.

As shown in FIG. 9, a stack of core sheets 18 and a stack of surface sheets 12, 14 are then placed on a conveyor 36. The core sheet 18 is preferably run through a vacuum bag 68 to minimize the consumption of the weight of the finished panels 10. The core sheet 18 is placed on a support 70 and a second surface sheet 14 is placed on top of the core sheet 18, a release sheet 72 is placed on top of the surface sheet 14, and a second set of surface sheets 12 are stacked up, i.e., a stack of assembled panels 10. The weight 62 (or a different pressure or vacuum bags) is applied to the assembled panels 10 and the adhesive 16 is cured. The honeycomb panels 10 are then removed and completed.

Referring back to FIGS. 1 and 2, it can now be appreciated that the core sheet 18 within the honeycomb panel 10 has a plurality of corrugated rigid resin ribbons 20 which preferably are fiber reinforced, that are arranged with these prepared and abraided apexes 24 abutted against each other and permanently bonded to each other with adhesive 47, which may be identical to the surface sheet adhesive 16 composition. All of the ribbons 20 are of identical height when it is intended to have the surface sheets 12, 14 parallel to each other. The complete honeycomb panel 10 is then ready for subsequent processing into a finished product. Examples of finished products include shelving, flooring, bulkheads, diving boards, docks, wall panels, partitions, ramps, pallets, swimming platforms, pontoon boat decks and the like. The preferred core sheet 18 has ribbons 20 taken from a panel 22 which has corrugation apexes 24 with flats that enhance preparation and bonding. The preferred cellular profile is generally hexagonal as seen in FIG. 2, but may also be generally square or of other configurations.

FIG. 9 illustrates a process wherein a prepared and abraided single corrugated panel 22 is sliced by the saw 64 to make individual core ribbons 20 that are subsequently bonded into a single layer core sheet 18 which may be flat or have single or multiple curvature of its surfaces.

Returning to FIG. 1, an integral flange 74 is shown on the surface sheet 12. The flange 74 is extendible around the entire periphery of the first surface sheet 12 providing hermetic sealing to the second surface sheet 14 during assembly and bonding of the finished honeycomb panel 10.

The new and improved honeycomb core sheet 18 and the new and improved completed honeycomb core panel 10 are usable in environments where conventional and known honeycombs will be environmentally attacked and fail. Specific examples are in building construction, sporting goods, vehicles, uses in agriculture and floating on water, uses in aggressive chemical environments, and uses in factories.

The new methods of fabricating the core sheet 18 and the finished core panels 10 can be effectively practiced by a single person, or a small business, or a large business. The materials to practice the method and build the finished products are available at most lumber yards. The finished core panel 10 is relatively low cost and is of high economic value.

Although other advantages may be found and realized and various modifications may be suggested by those versed in the art, be it understood that I embody within the scope of the patent warrant thereon, and such embodiments as reasonably and properly come within the scope of my contribution to the art.

I claim as my invention:

1. An improved honeycomb core comprising
  - (a) a plurality of corrugated ribbons or preformed rigid corrugated resin panels, said ribbons having mechanically abraided exterior apexes; and
  - (b) adhesive permanently bonding faces of said abraided apexes together.
2. The core of claim 1, in which each of said apexes has an abraided flat area with a transverse width which is at least  $\frac{1}{3}$  of the corrugation wavelength.
3. The core of claim 1, in which said resin is fiber reinforced.

of claim 3, in which said ribbons are of preformed polyester panels and in which the entire bonded apex is abraided.

4. The core of claim 1, in which every other ribbon is the mirror image of the immediately adjacent ribbons.

5. A honeycomb panel comprising

(a) a first surface sheet,

(b) a honeycomb core structure permanently secured to said surface sheet, said honeycomb core being a plurality of corrugated cells and secured first side to second side to second side and so on, said cell walls being formed by length ribbons of a preformed corrugated resin panel;

(c) mechanically abraided corrugation apexes on both sides of said ribbons; and

(d) adhesive permanently bonding opposing and abutted together faces of said apexes to each other.

6. A honeycomb panel of claim 6, in which the entire face of every corrugation apex is abraided.

7. A honeycomb panel, comprising

(a) a first surface sheet,

(b) a honeycomb core in between and spacing said surface sheets, said core being a plurality of cells formed from a rigid corrugated resin

(c) mechanically abraided exterior corrugation apexes on both sides of adjacent ribbons;

(d) adhesive permanently securing said abraided faces of said ribbons to each other; and

(e) adhesive permanently securing said surface sheets to said core.

8. A honeycomb panel of claim 8, in which said panel is substantially flat and said core being non-compressible and non-permeable to water.

9. A honeycomb panel of claim 8, in which said ribbons have mechanically abraided apexes adjoining to each other, said apexes being at least  $\frac{1}{3}$  of the width of each ribbon and each flat apex being abraided.

10. A method of making an improved honeycomb core comprising the steps of

(a) providing outer corrugation apexes on each panel in a plurality of rigid resin panels, in preparation for adhesive bonding;

(b) applying a layer of adhesive on the abraided

(c) placing said panels in a stack, with said adhesive on the apexes being in opposing face-to-face position to each other;

(d) pressing the panels together while curing the adhesive bonding the abraided apexes together to form a core block; and

(e) separating the core sheets off of an end of the core block.

11. A method of making an improved honeycomb core comprising the steps of

(a) providing a plurality of pieces of preformed rigid resin reinforced plastic panels;

(b) providing outer corrugation apexes on each panel in preparation for adhesive bonding and applying adhesive thereon;

(c) placing said panels in position against each other with the corrugation apexes of adjacent panels abutting against each other; and

(d) pressing the abutting abraided panels together and forming the improved honeycomb core from said rigid plastic panels.